

**AMENDMENTS TO THE CLAIMS**

This listing of Claims will replace all prior versions, and listings, of Claims in the application:

**Listing of Claims:**

Claim 1 (previously presented).      An aircraft brake heat pack brake disc in the form of a composite article comprising a core layer having a face portion and a wear layer attached to the face portion, wherein the core layer is a C-C composite article impregnated with a refractory carbide and the wear layer is a C-C composite article and has a density lower than the core layer.

Claim 2 (previously presented).      The aircraft brake heat pack brake disc as claimed in Claim 1, wherein the density of the core layer is in excess of 1.85 gcm<sup>-3</sup>.

Claim 3 (cancelled):

Claim 4 (cancelled):

Claim 5 (previously presented):      The aircraft brake heat pack brake disc as claimed in Claim 1, wherein the refractory carbide is silicon carbide or boron carbide.

Claim 6 (currently amended).      An aircraft brake heat pack comprising a brake disc in the form of a composite article comprising a core layer formed from C-C composite impregnated

with a refractory carbide, the core layer having a face portion to which is attached a carbide-free C-C wear layer having a density of 1.85 gcm<sup>-3</sup> or lower.

Claim 7 (previously presented).      The aircraft brake heat pack as claimed in Claim 6, wherein the refractory carbide is silicon carbide or boron carbide.

Claim 8 (previously presented).      The aircraft brake heat pack as claimed in Claim 6, wherein the density of the core layer is in excess of 1.85 gcm<sup>-3</sup>.

Claim 9 (previously presented).      The aircraft brake heat pack as claimed in Claim 8, wherein the density of the core layer is in the range of greater than 1.85 gcm<sup>-3</sup> to 2.95 gcm<sup>-3</sup>.

Claims 10-14 (cancelled).

Claim 15 (previously presented).      An aircraft wheel and brake assembly comprising brake discs, one or more of the brake discs having a core layer of density greater than 1.85 gcm<sup>-3</sup> and at least one wear layer attached to the core of density 1.85 gcm<sup>-3</sup> or lower, wherein the core layer comprises a C-C composite impregnated with refractory carbide.